

Minimization of Fuel Consumption of Natural Gas Compressor Stations with Similar and Dissimilar Turbo-Compressor Units

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Abstract

This paper studies and compares the results of a simple and fast heuristic method, Genetic Algorithm (GA) and the Exhaustive Search method (ES) concerning the minimization of fuel consumption of a natural gas compressor station (CS). The results, obtained for an input data set (natural gas flow rate of 150 million standard cubic meters per day (MMSCMD), suction pressure of 5.45 MPa and discharge pressure of 6.9 MPa), showed that for a CS with similar TC units, all the applied methods achieved the same solution (fuel consumption rate of 3.620 kg/s). By contrast, for a CS with dissimilar TC units, the GA and ES methods attained a lower fuel consumption rate (3.738 kg/s) compared to that obtained by the heuristic method (3.753 kg/s). The effect of changing the CS flow rate and CS suction and discharge pressures on optimal fuel consumption rate was also investigated. In the first case study, 100 MMSCMD (or 100%) increase in the flow rate, 8.3 bar (or 13%) increase in discharge pressure and 8.3 bar (or 14%) reduction in the suction pressure of the CS caused the optimal fuel consumption rate to increase by 2.41 kg/s (or 99%), 1.02 kg/s (or 33%) and 1.72 kg/s (or 60%), respectively. In the second case study, for the same changes of flow rate and discharge and suction pressures mentioned above, the optimal fuel consumption rate increased by 2.72 kg/s (or 112%), 1.03 kg/s (or 32%) and 1.71 kg/s (or 58%), respectively.

Keywords: Natural gas compressor station; Optimization; Fuel consumption; Genetic

Algorithm

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23 1. Introduction

24 Natural gas is increasingly used as a source of energy all over the world and the estimations show
25 that its worldwide consumption in 2030 will be twice as much as its present rate (Riva et al. 2006).

26 As natural gas travels through the transmission pipelines, the gas pressure drops due to both the
27 friction with the pipe walls and heat transfer to the surroundings. Therefore there is a need to
28 compensate the pressure by a number of compressor stations (CSs) located along the pipeline.

29 As an accepted rule of thumb, about 3 to 5% of the transmitted natural gas is consumed for gas
30 turbine drivers to generate power for the compressor stations which amounts to a huge cost
31 because of the large quantities of natural gas transmitted through extensive networks (Borraz-
32 Sanchez and Rios-Mercado 2009; Wu et al. 2000; Carter 1998). Therefore, even a slight
33 improvement in the performance of the gas transportation system can result in great savings.

34 The nonlinearity of the constraints and objective function, non-convexity of the compressors'
35 operating domain, possibility of on/off state for each turbo-compressor (TC) unit, and the
36 existence of local optima are some features of the CS optimization problem, which make this
37 problem very difficult to be solved using the classical mathematical methods (Borraz-Sanchez
38 and Rios-Mercado 2009; Wu et al. 2000; Chebouba et al. 2009; Rios-Mercado et al. 2006).

39 The heuristic methods are simple, fast, but unproven optimization tools that have originated from
40 past experiences. The complexity of natural gas system optimization problems highly encourages
41 the use of the heuristic methods (Borraz-Sánchez and Ríos-Mercado 2005). Davidson et al.
42 (2006) applied a heuristic method to minimize the investment cost of a natural gas distribution
43 network. Carter (1996) has described the commonly used heuristic methods for the optimization
44 of fuel consumption in the CSs.

45 With the ability of setting the control parameters, the evolutionary Genetic Algorithm (GA) is
46 considered as an efficient and powerful optimization tool. Nguyen et al. (2008), and Nguyen and
47 Chan (2006) have given a review of the research works that have used GA method.
48 The present paper is concerned with the minimization of fuel consumption of CSs. The on/off
49 state for each of the TC units existing in the CS, and also the rate of flow passing through each of
50 the running TC units were considered as the decision variables of the optimization problem. The
51 Pataveh CS with similar TC units located along the third Iranian gas transmission pipeline
52 (IGAT) and the same CS but with the assumption of having dissimilar TC units constitute the
53 case studies of the present paper. A heuristic method based on the same utilization value
54 (Utilization is defined as the ratio of the rate of flow passing through a TC unit to the maximum
55 permissible flow rate) for all running units, and Genetic Algorithm (GA) were applied as the
56 optimization methods and their results and computation times were compared. Furthermore, the
57 results of the heuristic and GA methods were checked through comparison with the results
58 obtained by the Exhaustive Search method (ES). Finally, the impact of changing the values of
59 important input parameters of the studied CS on optimal fuel consumption rate was investigated.
60 The rest of this paper is organized as follows. The modeling and governing equations of a typical
61 CS are described in section 2. The description of the optimization problem as well as a brief
62 introduction to the optimization methods are presented in section 3. The case studies are
63 introduced in section 4, and the results are presented and discussed in section 5.

64

65 **2. Modeling and the governing equations**

66 Gas transmission networks are composed of pipelines and CSs as the main components. A
67 schematic diagram of a typical gas network, including a number of CSs and pipelines, is shown
68 in Figure (1). As natural gas travels through the transmission pipelines from a supply point (point

79 A in Figure (1)) to some delivery points (points B and C in Figure (1)), its pressure drops, mainly
 80 due to the friction with pipe walls. This pressure drop should be compensated by a number of
 81 CSs located along the pipelines to flow natural gas to the delivery points at required values of
 82 pressure and volume flow rate.

83 A schematic diagram of a CS located between two pipelines, with a number of TC units in
 84 parallel, is shown in Figure (2). As shown in Figure (2), each TC unit consists of a natural gas
 85 compressor (responsible for compensating the pressure drop), and a gas turbine (also called
 86 turbine engine) as the driver of the natural gas compressor. The gas turbine is a double-shaft
 87 turbine composed of air compressor, combustion chamber, and low and high pressure (LP, HP)
 88 turbines, which will be discussed in detail in section (2.2). The "natural gas compressor" is
 89 henceforth shortened to "compressor". The modeling and governing equations of typical CS
 90 components are described as follows:

91 **2.1. Natural gas compressor (pipeline compressor)**

92 Equations (1)-(9) show the governing equations of gas flow passing through a typical
 93 compressor:

94 The ratio of compressor isentropic head to the square of rotational speed:

$$95 \quad \frac{H}{S^2} = b_1 + b_2 \left(\frac{Q_{ac}}{S} \right) + b_3 \left(\frac{Q_{ac}}{S} \right)^2 \quad (1)$$

96 The compressor isentropic efficiency:

$$97 \quad \eta_{c,is} = b_4 + b_5 \left(\frac{Q_{ac}}{S} \right) + b_6 \left(\frac{Q_{ac}}{S} \right)^2 \quad (2)$$

98 In equations (1) and (2) as empirical equations proposed by Odom (1990), and Percell and Ryan
 99 (1987), b_1 to b_6 are the constant coefficients obtained from the compressor operating map.

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90 The compressor isentropic head in terms of compressor pressure ratio:

$$H = \frac{Z_s RT_s}{\sigma} \left[\left(\frac{p_d}{p_s} \right)^\sigma - 1 \right] \quad (3)$$

91 The compressibility factor (Z_s) (Mohring et al. 2004):

$$Z_s = 1 + 0.257(p_s / p_c) - 0.533(p_s / p_c)(T_c / T_s) \quad (4)$$

92 The compressor discharge temperature after gas compression:

$$T_d = T_s + \frac{T_s}{\eta_{c,is}} \left[\left(\frac{p_d}{p_s} \right)^\sigma - 1 \right] \quad (5)$$

93 The compressor power consumption in terms of compressor mechanical efficiency ($\eta_{c,m}$):

$$Power_{shaft} = \frac{H \dot{m}_d}{\eta_{c,is} \cdot \eta_{c,m}} \quad (6)$$

94 The required fuel mass flow rate in gas turbine for running the compressor:

$$\dot{m}_f = \frac{Power_{shaft}}{LHV \cdot \eta_{th,gasturbine}} \quad (7)$$

95 The mass balance in a TC unit (as shown in Figure (2)):

$$\dot{m}_f + \dot{m}_d = \dot{m}_s \quad (8)$$

96 In the above equations, subscriptions s and d indicate compressor suction and discharge points.

97 The actual volumetric flow rate passing through a compressor as a function of mass flow rate,

98 pressure and temperature, is obtained from:

$$Q_{ac} = \frac{\dot{m}_d Z_s RT_s}{p_s} \quad (9)$$

99 2.2. Gas turbine (Driver)

100 Generally, two-shaft gas turbines are used in gas pipeline applications because of their

101 operational flexibility (Cohen et al. 1987). Part of Figure (2) confined in a dashed box shows a

102 schematic diagram of a two-shaft gas turbine. A gas turbine (including air compressor,
103 combustion chamber, and low and high pressure turbines) provides power to run the air
104 compressor by the high pressure turbine and to run the pipeline compressor by the low pressure
105 turbine (also called power turbine).

106 The technical specifications of turbine engines are generally associated with their ISO conditions
107 and base load operation. These specifications include maximum power and efficiency
108 ($Power_A$ and $\eta_{th,A}$) and the output shaft's rotational speed (S_A) at which maximum power and
109 efficiency are produced. In most cases though, the turbine operates outside these ISO, base load
110 and optimal conditions; therefore, some corrections are required for estimating the turbine
111 engine's overall performance.

112 **Correction for the ambient temperature**

113 Changes of ambient temperature alter maximum power and efficiency as well as the output
114 shaft's rotational speed at which power and efficiency attain their maximum values (Kurz and
115 Ohanian 2003; Santos 1997). The following equations were used to take these effects into
116 consideration:

$$\frac{Power_A}{Power_B} = f_1 \left(\frac{T_{ambient}(K)}{T_{iso}(K)} \right) \quad (10)$$

$$\frac{\eta_{th,A}}{\eta_{th,B}} = f_2 \left(\frac{T_{ambient}(K)}{T_{iso}(K)} \right) \quad (11)$$

$$\frac{S_B}{S_A} = f_3 \left(\frac{T_{ambient}(K)}{T_{iso}(K)} \right) \quad (12)$$

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۱۲۰ **Correction due to the part load operation**

۱۲۱ The turbine part load operation decreases the maximum values of power and efficiency, and also
۱۲۲ decreases the rotational speed at which power and efficiency attain their maximum values (Kurz
۱۲۳ and Ohanian 2003; Walsh and Fletcher 2004). Equations which consider these effects are:

$$\frac{\eta_{th,C}}{\eta_{th,B}} = f_4 \left(\frac{Power_C}{Power_B} \right) \quad (13)$$

$$\frac{S_C}{S_B} = f_5 \left(\frac{Power_C}{Power_B} \right) \quad (14)$$

۱۲۴ **Correction due to the operation out of design rotational speed**

۱۲۵ For any operating condition of a gas generator (part of the gas turbine which includes air
۱۲۶ compressor, combustion chamber and HP turbine as was shown in Figure (2)), there is a
۱۲۷ rotational speed at which the output shaft power and efficiency values are the highest. If the
۱۲۸ power turbine deviates from this speed, the power and efficiency decrease with the same
۱۲۹ proportions (Kurz and Ohanian 2003). Equations (15) and (16) were proposed by Kurz and
۱۳۰ Ohanian (2003), and Kurz and Brun (2009) to take this effect into consideration:

$$\frac{Power_{shaft}}{Power_C} = 2 \left(\frac{S}{S_C} \right) - \left(\frac{S}{S_C} \right)^2 \quad (15)$$

$$\frac{\eta_{th,gasturbine}}{\eta_{th,C}} = \frac{Power_{shaft}}{Power_C} \quad (16)$$

۱۳۱ Where, $Power_{shaft}$ and S represent the power and rotational speed required by the pipeline
۱۳۲ compressor, respectively.

۱۳۳ The set of nonlinear equations, including Equations (1)-(16) for each TC unit, was solved using
۱۳۴ Newton-Raphson method in this paper. Newton-Raphson method is a widely used method for

135 solving a set of nonlinear equations which has the advantage of converging quadratically (Gerald
 136 and Wheatley 1999).

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138 **3. Optimization**

139 For the specified rate of flow passing through the CS, and the specified suction and discharge
 140 pressures of the CS, the optimization problem in this paper is supposed to determine the on/off
 141 state for each TC unit in the CS and the rate of flow allocated to each of the active units, so that
 142 the fuel consumption rate of the CS is minimized.

143 The optimization problem is mathematically expressed by Equation (17) (Carter 1996), where n
 144 is the number of TC units existing in the CS, J_i is an integer variable with the value of 1/0 for
 145 the on/off state of unit i, and Q_i is the volume flow rate of natural gas that enters unit i.

$$\text{Minimize} \left[\sum_{i=1}^n J_i \cdot \dot{m}_{f_i}(Q_i) \right] ; J_i \in \{0,1\} \quad (17)$$

$$\text{Constraints} \begin{cases} \sum_{i=1}^n J_i Q_i = Q_{station} \\ \text{if } J_i = 1 \rightarrow Q_{i,min} \leq Q_i \leq Q_{i,max} \end{cases}$$

146 The first constraint in Equation (17) states that the sum of the input flow rates of the running
 147 units must be equal to the inlet flow rate ($Q_{station}$) of the CS. The second constraint is related to
 148 the feasible operating domain of each unit, and states that the amount of flow which enters each
 149 running unit must be between certain minimum and maximum values.

150 The feasible operating domain of a typical compressor (bounded by maximum and minimum
 151 speed, surge and stonewall lines) as well as the compressor minimum and maximum permissible
 152 volume flow rates for an arbitrary isentropic head are shown in Figure (3).

103 The optimization methods applied in this paper (which their codes were developed by the authors
104 at the Energy Systems Improvement Laboratory (ESIL)) are briefly described as follows:

105 **Heuristic optimization method based on equal utilization**

106 The heuristic method applied in the present paper for the minimization of the CS fuel consumption
107 rate is based on the assumption of the same utilization value for all running TC units (Carter 1996).

108 This method is henceforth shortened to the "heuristic method". Utilization is defined as the ratio of the
109 rate of flow passing through a TC unit to the maximum permissible flow rate through that TC unit

110 ($\alpha = Q_i / Q_{i,max}$). It should be noted that equal utilization values implies equal flow rates for similar
111 units (since similar units have identical maximum permissible flow rates), however, this notion doesn't
112 hold true for dissimilar units.

113 Figure (4) schematically shows the flow chart of steps to be followed to obtain the fuel
114 consumption rate based on the heuristic method, for a specified combination of on/off states of
115 the units (i.e., for a specified J). These steps are followed for each possible combination; and the
116 combination with the minimum fuel consumption rate is selected as the optimal solution.

117 **Genetic optimization Algorithm (GA)**

118 Evolutionary algorithms are random search methods that mimic the natural evolution. These
119 algorithms start with a population of possible solutions and repeatedly generate a new population
120 from the last one based on the survival of fitter solutions, with the hope of finding solutions with
121 better objective functions. The flow chart of GA steps (as an evolutionary algorithm) is shown in
122 Figure (5). These steps are also briefly described below (Gen and Cheng 2000; El-Mahdy et al.
123 2010; Hu and Fang 2012):

124 *Formation of the first population*

125 The formation of the first population (generation) is the first step in GA procedure. A population
126 consists of a number of chromosomes (individuals), each a string of coded bits (genes), which

187 represents a single solution of the problem under study. The first population is created by
188 randomly choosing the binary value of 0 or 1 for each bit.

189 *Selection process and the mating pool*

190 The next step is to select some individuals to produce the offspring (children) and establish the
191 new generation. In the selection process, individuals with better objective functions have a
192 greater chance of being selected. A collection of selected individuals makes up the mating pool.

193 *Crossover and mutation*

194 After parents are randomly chosen from the selected individuals in the mating pool, they undergo
195 the crossover procedure in order to produce offspring. Crossover produces offspring that inherit
196 their genes from both parents. Then, mutation is applied on the children, which alters the initial
197 values of some of their genes. Mutation makes it possible for the children to have some different
198 gene values than their parents.

199 *Elitism*

200 Elitism is a method which copies the best chromosome (or a few best chromosomes) of every
201 generation to the next generation without any change.

202 The Roulette-Wheel selection, single point crossover, and the uniform mutation were the GA
203 operators used in this paper. Furthermore, the best chromosome of every generation was copied
204 to the next generation without any change. Values of 80%, 5% and 100 were chosen for
205 crossover rate, mutation rate and population size, respectively. Unchanged best solution for a
206 sequence of 50 consecutive generations was considered as the convergence criterion for the GA.

207 **Exhaustive Search method (ES)**

208 ES method computes the objective functions for all possible solutions, and selects the solution
209 with the best objective function as the optimal solution. Since ES method certainly reaches the
210 global optimal solution, it will be the preferred method of optimization if it can solve the problem

۲۰۱ under study within a desired run time. However, except for simple cases (cases which do not
۲۰۲ include many kinds and numbers of TC units, such as one studied in this paper), and when the
۲۰۳ step size for continuous decision variables (i.e. volume flow rate in the problem studied in this
۲۰۴ paper) is coarse, ES method requires unreasonable computing time. It should be noted that ES
۲۰۵ method has been applied here just as a guide for checking the accuracy of heuristic and GA
۲۰۶ optimization results.

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۲۰۸ **4. Case studies**

۲۰۹ Two case studies were investigated in the present paper.

۲۱۰ The Pataveh CS with six parallel A-type TC units, located along the third Iranian gas
۲۱۱ transmission pipeline at a distance of 423 km from the starting point, was considered in the first
۲۱۲ case study.

۲۱۳ In the second case study, the Pataveh CS was assumed with three A-type and three B-type TC
۲۱۴ units (total of six units).

۲۱۵ The constant coefficients of A-type and B-type compressors are shown in Table (1). Parts of the
۲۱۶ input data which were the same for both case studies have been listed in Table (2).

۲۱۷ Constant coefficients of four second-order polynomial equations ($a_1 x^2 + a_2 x + a_3$), which were
۲۱۸ obtained (by curve fitting) for the functions f_1 to f_3 and f_5 in Equations (10)-(12) and (14), are
۲۱۹ shown in Table (3). Also, a logarithmic function was obtained for f_4 ($f_4 = 0.2457 \ln(x) + 1$). It

۲۲۰ is worth mentioning that the above functions have been verified by Sanaye and Mahmoudimehr
۲۲۱ (2012a).

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۲۲۴ **5. Discussion and Results**

۲۲۵ **5.1. Results of the first case study (CS with similar TC units)**

۲۲۶ In the first case study, both the GA and heuristic methods achieved the same solution, which was
۲۲۷ also verified by the ES method. The optimum values of decision variables, obtained by the
۲۲۸ optimization procedures, were shown in Table (4). Furthermore, some of computed operating
۲۲۹ parameters at the optimum point as well as optimum value of the objective function (total fuel
۲۳۰ consumption rate) were listed in Table (5). Tables (4) and (5) indicate that four out of six TC
۲۳۱ units in a CS should operate identically (i.e., with equal values of flow rate, utilization, speed,
۲۳۲ required power, etc.) in order to achieve the optimal solution. This finding indicated that the
۲۳۳ assumption of 'same utilization' that was used by the heuristic method for all the running TC
۲۳۴ units led to optimal solution and that the GA or ES method could not achieve a better solution.
۲۳۵ The flow rate and suction and discharge pressures are some of the operating parameters that
۲۳۶ significantly affect the fuel consumption rate of a CS. To study the effect of each of the
۲۳۷ aforementioned parameters on the optimal fuel consumption rate, the values of all the other
۲۳۸ parameters in the input data were kept constant (at the values listed in Tables 1-3) and the optimal
۲۳۹ fuel consumption rate was computed for different values of the investigated parameter.
۲۴۰ Figs. 6-8 show the changes of optimal fuel consumption rate with CS flow rate, suction pressure
۲۴۱ and discharge pressure, respectively. According to these figures, an increase in the flow rate or
۲۴۲ discharge pressure of the CS or a reduction in its suction pressure led to an increase in the optimal
۲۴۳ fuel consumption rate. This outcome was due to the fact that each of the aforementioned changes
۲۴۴ increased the required power consumption of the CS according to Equations (3) and (6). As is
۲۴۵ illustrated in Figs. 6-8, with the variations of CS flow rate, suction pressure and discharge pressure
۲۴۶ from 100 to 200 MMSCMD, 5.03 to 5.86 MPa and 6.48 to 7.31 MPa, respectively, the optimal

fuel consumption rate correspondingly varied from 2.430 to 4.836 kg/s, 4.587 to 2.869 kg/s and 3.099 to 4.115 kg/s.

5.2. Results of the second case study (CS with dissimilar TC units)

The optimization results obtained by the heuristic, GA and ES methods are presented and compared in Tables (6) and (7). As is shown in these Tables, all the applied methods predicted the same number of A- and B-type active units (three A-type units and one B-type unit). However, GA attained a lower fuel consumption rate (3.738 kg/s) compared to that obtained by the heuristic method (3.753 kg/s). It should be noted that the ES results confirmed the validity of GA results. The reason for the discrepancy in fuel consumption rates obtained by the heuristic and GA methods is that, in contrast to the heuristic method which obtained the same utilization value (91.4%) for all the running units, the GA method predicted different values of utilization for dissimilar units (95% for A-type units and 75.7% for B-type units). This finding indicates that although the assumption made by the heuristic method (equal utilization values for all the running units) was accurate for the CS with similar units, it did not lead to optimal solution for the CS with dissimilar units.

The effects of the CS operating parameters on optimal fuel consumption rate were also investigated for the second case study. As is illustrated in Figs. 6-8, in the second case study, with the variations of CS flow rate, suction pressure and discharge pressure from 100 to 200 MMSCMD, 5.03 to 5.86 MPa and 6.48 to 7.31 MPa, respectively, the optimal fuel consumption rate correspondingly varied from 2.430 to 5.152 kg/s, 4.690 to 2.978 kg/s and 3.225 to 4.251 kg/s. The average run times of heuristic, GA and ES methods on an Intel (R) Core (TM) i5 2.53 GHz processor were about 0.02, 34.6 and 274.8 seconds, respectively.

۲۷۱ **6. Conclusions**

۲۷۲ The minimization of fuel consumption by natural gas compressor stations (CSs) was studied in
۲۷۳ this paper. A heuristic method (assuming the same utilization value for all the running turbo-
۲۷۴ compressor (TC) units), the Genetic Algorithm (GA) and the Exhaustive Search method (ES)
۲۷۵ were employed as optimization methods.

۲۷۶ For the CS with similar TC units, the optimization methods applied in this paper achieved the
۲۷۷ same solution (a fuel consumption rate of 3.620 kg/s). In this case, the simple and fast heuristic
۲۷۸ method achieved a global optimal solution, and it was unnecessary to use other methods with
۲۷۹ higher degrees of complexity and longer computation times.

۲۸۰ By contrast, for the CS with dissimilar units, the assumption made by the heuristic method (equal
۲۸۱ utilizations for all the running units regardless of their type) did not lead to optimal solution. In
۲۸۲ this case, by considering different utilization values for dissimilar TC units (95% for A-type units
۲۸۳ and 75.7% for B-type units), the GA method could attain a lower fuel consumption rate (3.738
۲۸۴ kg/s) compared to that obtained by the heuristic method (3.753 kg/s). The mentioned heuristic
۲۸۵ method assumed the same utilization value of 91.4% for all the running units.

۲۸۶ Finally, it was observed that an increase in CS flow rate and discharge pressure and a reduction in
۲۸۷ CS suction pressure results in an increase in the optimal fuel consumption rate.

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۲۹۵ **Abbreviations**

۲۹۶ CS = Compressor station

۲۹۷ ES = Exhaustive search method

۲۹۸ MMSCMD = Million standard cubic meters per day

۲۹۹ TC = Turbo-compressor

۳۰۰ **Notation**

۳۰۱ b_1 to b_6 = Constant values for modeling a compressor

۳۰۲ f_1 to f_6 = Functions considered for modeling a two-shaft gas turbine

۳۰۳ H = Isentropic head (MJ/kg)

۳۰۴ J_i = An integer variable with the value of 1/0 for the on/off state of unit i

۳۰۵ LHV = Lower heating value of natural gas (MJ/kg)

۳۰۶ \dot{m}_d = Mass flow rate of natural gas passing through the compressor (kg/s)

۳۰۷ \dot{m}_f = Mass flow rate of fuel (natural gas) consumed by the gas turbine engine (kg/s)

۳۰۸ $\dot{m}_s = \dot{m}_f + \dot{m}_d$ = Mass flow rate of natural gas for one turbo-compressor unit (kg/s)

۳۰۹ p_c = Natural gas critical pressure (MPa)

۳۱۰ p_d = Discharge pressure (MPa)

۳۱۱ p_s = Suction pressure (MPa)

۳۱۲ $Power_A$ = Gas turbine maximum power output under ISO conditions and the base load operation (MW)

۳۱۳ $Power_B$ = Gas turbine maximum power output at a typical ambient temperature and the base load

۳۱۴ operation (MW)

۳۱۵ $Power_C$ = Gas turbine maximum power output at a typical ambient temperature and a part load

۳۱۶ operation (MW)

- 317 $Power_{shaft}$ = Compressor power consumption (MW)
- 318 Q_{ac} = Actual volumetric flow rate of natural gas passing through a compressor (m^3/s)
- 319 Q_i = Actual volumetric flow rate of natural gas that enters unit i (m^3/s)
- 320 $Q_{i,max}$ = Maximum permissible flow rate of natural gas that enters unit i (m^3/s)
- 321 $Q_{i,min}$ = Minimum permissible flow rate of natural gas that enters unit i (m^3/s)
- 322 $Q_{station}$ = CS actual volumetric flow rate (m^3/s)
- 323 R = Gas constant (MJ/kg.K)
- 324 S = Rotational speed (rpm)
- 325 S_A = Rotational speed at which maximum power and efficiency are produced under ISO conditions and the base load operation (rpm)
- 326 S_B = Rotational speed at which maximum power and efficiency are produced under a typical ambient temperature and the base load operation (rpm)
- 327 S_C = Rotational speed at which maximum power and efficiency are produced under a typical ambient temperature and a part load operation, (rpm)
- 328 $T_{ambinet}$ = Ambient temperature (K)
- 329 T_c = Natural gas critical temperature (K)
- 330 T_d = Discharge temperature (K)
- 331 T_{iso} = Temperature at ISO conditions (K)
- 332 T_s = Suction temperature (K)
- 333 Z_s = Compressibility factor at the suction side
- 334 α = Utilization

- 338 $\eta_{c,is}$ = Compressor isentropic efficiency
- 339 $\eta_{c,m}$ = Compressor mechanical efficiency
- 340 $\eta_{th,A}$ = Gas turbine maximum thermal efficiency under ISO conditions and the base load operation
- 341 $\eta_{th,B}$ = Gas turbine maximum thermal efficiency at a typical ambient temperature and the base
- 342 load operation
- 343 $\eta_{th,C}$ = Gas turbine maximum thermal efficiency at a typical ambient temperature and a part load
- 344 operation
- 345 $\eta_{th,gasturbine}$ = Gas turbine thermal efficiency
- 346 σ = Isentropic exponent
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361 **References**

- 362 Borraz-Sánchez, C. and Ríos-Mercado, R.Z. (2005). "Hybrid metaheuristics." *A hybrid meta-*
363 *heuristic approach for natural gas pipeline network optimization*, M. J. Blesa, C. Blum, A. Roli,
364 and M. Sampels, eds., Springer, Berlin, Germany, 54-65.
- 365 Borraz-Sánchez, C., and Ríos-Mercado, R.Z. (2009). "Improving the operation of pipeline
366 systems on cyclic structures by tabu search." *Comput. Chem. Eng.*, 33(1), 58-64.
- 367 Carter, R.G. (1996). "Compressor station optimization: computational accuracy and speed." *28th*
368 *Annual Meeting Pipeline Simulation Interest Group (PSIG)*, San Francisco, California, USA,
369 (23-25 Oct).
- 370 Carter, R.G. (1998). "Pipeline Optimization: Dynamic Programming after 30 Years." *30th*
371 *Annual Meeting Pipeline Simulation Interest Group (PSIG)*, Denver, Colorado, USA, (28-30
372 Oct).
- 373 Chebouba, A., Yalaoui, F., Smati, A., Amodeo, L., Younsi, K., and Tairi, A. (2009).
374 "Optimization of natural gas pipeline transportation using ant colony optimization." *Comput.*
375 *Oper. Res.*, 36(6), 1916-1923.
- 376 Cohen, H., Rogers, G.F.C., and Saravanamuttoo, H.I.H. (1987) "Gas Turbine Theory." Longman
377 Publication Group, London, 336-369.
- 378 Davidson, R.A., Lembo, A.J., Ma, J., Nozick, L.K., and O'Rourke, T.D. (2006). "Optimization of
379 investments in natural gas distribution networks." *J. Energy Eng.*, 132 (2), 52-60.
- 380 El-Mahdy, O.F.M., Ahmed, M.E.H., and Metwalli, S. (2010). "Computer aided optimization of
381 natural gas pipe networks using genetic algorithm." *Appl. Soft Comput.*, 10(4), 1141-1150.

- 382 Gen, M., and Cheng, R. (2000). “*Genetic algorithms and engineering optimization.*” John Wiley
383 & Sons, New York.
- 384 Gerald, C.F., and Wheatley, P.O., (1999) “*Applied numerical analysis.*” Addison Wesley
385 Longman Inc, Boston, 174-178.
- 386 Hu, W., and Fang, Y. (2012). “Multi-model parameters identification for main steam temperature
387 of ultra-supercritical units using improved genetic algorithm.” *J. Energy Eng.*, in press.
- 388 Kurz, R., and Brun, K. (2009). “Degradation of gas turbine performance in natural gas service.”
389 *J. Nat. Gas Sci. Eng.*, 1(3), 95-102.
- 390 Kurz, R., and Ohanian, S. (2003). “Modeling turbomachinery in pipeline simulations.” *35th*
391 *Annual Meeting Pipeline Simulation Interest Group (PSIG)*, Bern, Switzerland, (15-17 Oct).
- 392 Mohring, J., Hoffmann, J., Halfmann, T., Zemitis, A., and Basso, G. (2004). “Automated model
393 reduction of complex gas pipeline networks”. *36th Annual Meeting Pipeline Simulation Interest*
394 *Group (PSIG)*, Palm Springs, California, USA, (20-22 Oct).
- 395 Nguyen, H.H., and Chan, C.W. (2006). “Applications of artificial intelligence for optimization of
396 compressor scheduling.” *Eng. Appl. Artif. Intel.*, 19(2), 113-126.
- 397 Nguyen, H.H., Uraikul, V., Chan, C.W., and Tontiwachwuthikul, P. (2008). “A comparison of
398 automation techniques for optimization of compressor scheduling.” *Adv. Eng. Softw.*, 39(3), 178-
399 188.
- 400 Odom, F.M. (1990). “Tutorial on modeling of gas turbine driven centrifugal compressors.” *22nd*
401 *Annual Meeting Pipeline Simulation Interest Group (PSIG)*, Baltimore, Maryland, USA (18-19
402 Oct).

- ε·³ Percell, P.B., and Ryan, M. J. (1987). “Steady-state optimization of gas pipeline network
ε·⁴ operation.” *19th Annual Meeting Pipeline Simulation Interest Group (PSIG)*, Tulsa, Oklahoma,
ε·⁵ USA, (22-23 Oct).
- ε·⁶ Rios-Mercado, R.Z., Kim, S., and Boyd, E.A. (2006). “Efficient operation of natural gas
ε·⁷ transmission systems: a network-based heuristic for cyclic structures.” *Comput. Oper. Res.*,
ε·⁸ 33(8), 2323-2351.
- ε·⁹ Riva, A., Angelosante, S.D., and Trebeschi, C. (2006). “Natural gas and the environmental
ε·¹⁰ results of lifecycle assessment.” *Energy.*, 31(1), 138-148.
- ε·¹¹ Sanaye, S., and Mahmoudimehr, J. (2012a). “Technical assessment of isothermal and non-
ε·¹² isothermal modelings of natural gas pipeline operational conditions.” *Oil Gas Sci. Technol.*,
ε·¹³ 67(3), 435-449.
- ε·¹⁴ Sanaye, S., and Mahmoudimehr, J. (2012b). “Minimization of fuel consumption in cyclic and
ε·¹⁵ non-cyclic natural gas transmission networks: assessment of genetic algorithm optimization
ε·¹⁶ method as an alternative to non-sequential dynamic programming.” *J. Taiwan Inst. Chem. Eng.*,
ε·¹⁷ 43(6), 904-917.
- ε·¹⁸ Santos, S.P.D. (1997). “Transient analysis a must in gas pipeline design.” *29th Annual Meeting
ε·¹⁹ Pipeline Simulation Interest Group (PSIG)*, Tucson ,Arizona, USA, (15-17 Oct).
- ε·²⁰ Walsh, P. P., and Fletcher, P. (2004). “*Gas turbine performance.*” Blackwell Science Ltd,
ε·²¹ Oxford, UK, 383-443.
- ε·²² Wu, S., Rios-Mercado, R.Z., Boyd, E.A., and Scott, L. R. (2000). “Model relaxations for the fuel
ε·²³ cost minimization of steady-state gas pipeline networks.” *Math. Comput. Model.* , 31 (2-3) , 197-
ε·²⁴ 220.

٤٢٥ **Table captions:**

٤٢٦ **Table 1.** Constant coefficients for the A- and B-type compressors

٤٢٧ **Table 2.** Input data for both case studies

٤٢٨ **Table 3.** Constant coefficients of the functions f_1 to f_3 and f_5 in Equations (10)-(12) and (14)

٤٢٩ **Table 4.** Optimum value of decision variables obtained by the heuristic, GA and ES methods in
٤٣٠ the first case study

٤٣١ **Table 5.** Computed operating parameters at the optimum point obtained by the heuristic, GA and
٤٣٢ ES methods in the first case study

٤٣٣ **Table 6.** Optimum value of decision variables obtained by each of the heuristic, GA and ES
٤٣٤ methods in the second case study

٤٣٥ **Table 7.** Computed operating parameters at the optimum point obtained by each of the heuristic,
٤٣٦ GA and ES methods in the second case study

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Table 1. Constant coefficients of the A- and B-type compressors

	A-type compressor	B-type compressor
b_1	4.8089×10^{-4}	8.294×10^{-4}
b_2	1.2287	1.897
b_3	-1.1086×10^3	-2.532×10^3
b_4	-1.4005	13.929
b_5	2.1932×10^5	2.54×10^5
b_6	-1.4016×10^8	-2.289×10^8
$(Q/S)_{min} \left(\frac{m^3/s}{rev/min} \right)$	6.15×10^{-4}	3.76×10^{-4}
$(Q/S)_{max} \left(\frac{m^3/s}{rev/min} \right)$	11.78×10^{-4}	8.53×10^{-4}

Table 2. Input data for both case studies

Rate of flow passing through the CS (MMSCMD)	150
Suction pressure (p_s) (MPa)	5.45
Discharge pressure (p_d) (MPa)	6.89
Suction temperature (T_s) (K)	288.15
Ambient temperature ($T_{ambient}$) (K)	288.15
Maximum allowable driver speed (S_{max}) (rpm)	7700
Minimum allowable driver speed (S_{min}) (rpm)	4500
Turbine output power at the design point (MW)	25.4
Turbine efficiency at the design point (%)	35.1
Turbine speed at the design point (rpm)	7350

402 **Table 3.** Constant coefficients of the functions f_1 to f_3 and f_5 in Equations (10)-(12) and (14)

function	a_1	a_2	a_3
f_1	-4.3115	6.6618	-1.3618
f_2	-2.4918	4.4951	-1.0074
f_3	-0.4275	0.6710	0.7566
f_5	-0.397	1.0165	0.3777

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 404 **Table 4.** Optimum value of decision variables obtained by the heuristic, GA and ES methods in
 405
 406 the first case study

	Optimum on/off state	Optimum TC flow rate (MMSCMD)
TC unit1	On	37.5
TC unit2	On	37.5
TC unit3	On	37.5
TC unit4	On	37.5
TC unit5	Off	0
TC unit6	Off	0

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ε66 **Table 5.** Computed operating parameters at the optimum point obtained by the heuristic, GA and
 ε67 ES methods in the first case study

	Utilization (%)	Compressor speed (rpm)	Compressor required power (MW)	Fuel consumption rate (kg/s)
TC unit1	84.4	7035	12.5	0.905
TC unit2	84.4	7035	12.5	0.905
TC unit3	84.4	7035	12.5	0.905
TC unit4	84.4	7035	12.5	0.905
TC unit5	0	0	0	0
TC unit6	0	0	0	0
Total fuel consumption rate (kg/s)				3.620

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ε69 **Table 6.** Optimum value of decision variables obtained by each of the heuristic, GA and ES
 ε70 methods in the second case study

	Heuristic method		GA and ES methods	
	Optimum on/off state	Optimum TC flow rate (MMSCMD)	Optimum on/off state	Optimum TC flow rate (MMSCMD)
TC unit 1 (A-type)	On	40.6	On	42.2
TC unit 2 (A-type)	On	40.6	On	42.2
TC unit 3 (A-type)	On	40.6	On	42.2
TC unit 4 (B-type)	On	28.2	On	23.4
TC unit 5 (B-type)	Off	0	Off	0
TC unit 6 (B-type)	Off	0	Off	0

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٤٧٤ **Table 7.** Computed operating parameters at the optimum point obtained by each of the heuristic,
 ٤٧٥ GA and ES methods in the second case study

	Heuristic method				GA and ES methods				
	Utilization (%)	Speed (rpm)	Compressor Power (MW)	Fuel consumption rate (kg/s)	Utilization (%)	Speed (rpm)	Compressor Power (MW)	Fuel consumption rate (kg/s)	
TC unit 1 (A-type)	91.4	7316	14.1	0.982	95.0	7469	15	1.026	
TC unit 2 (A-type)	91.4	7316	14.1	0.982	95.0	7469	15	1.026	
TC unit 3 (A-type)	91.4	7316	14.1	0.982	95.0	7469	15	1.026	
TC unit 4 (B-type)	91.4	6542	10.6	0.806	75.7	5930	8	0.66	
TC unit 5 (B-type)	0	0	0	0	0	0	0	0	
TC unit 6 (B-type)	0	0	0	0	0	0	0	0	
Total fuel consumption rate (kg/s)				3.753	Total fuel consumption rate (kg/s)				3.738

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000 **Figure captions:**

001 **Fig.1.** Schematic diagram of a typical natural gas transmission network

002 **Fig.2.** Schematic diagram of a typical natural gas compressor station

003 **Fig.3.** Feasible operating domain of a typical compressor

004 **Fig.4.** Flow chart of the heuristic method

005 **Fig.5** Flow chart of GA

006 **Fig.6.** Optimal value of fuel consumption mass flow rate as a function of compressor station

007 volume flow rate

008 **Fig.7.** Optimal value of fuel consumption mass flow rate as a function of compressor station

009 suction pressure

010 **Fig.8.** Optimal value of fuel consumption mass flow rate as a function of compressor station

011 discharge pressure

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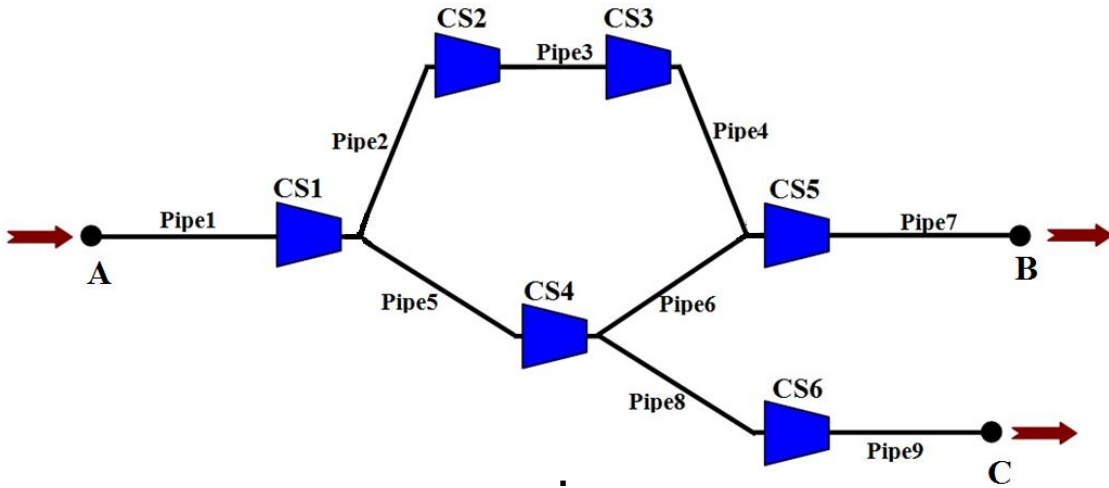
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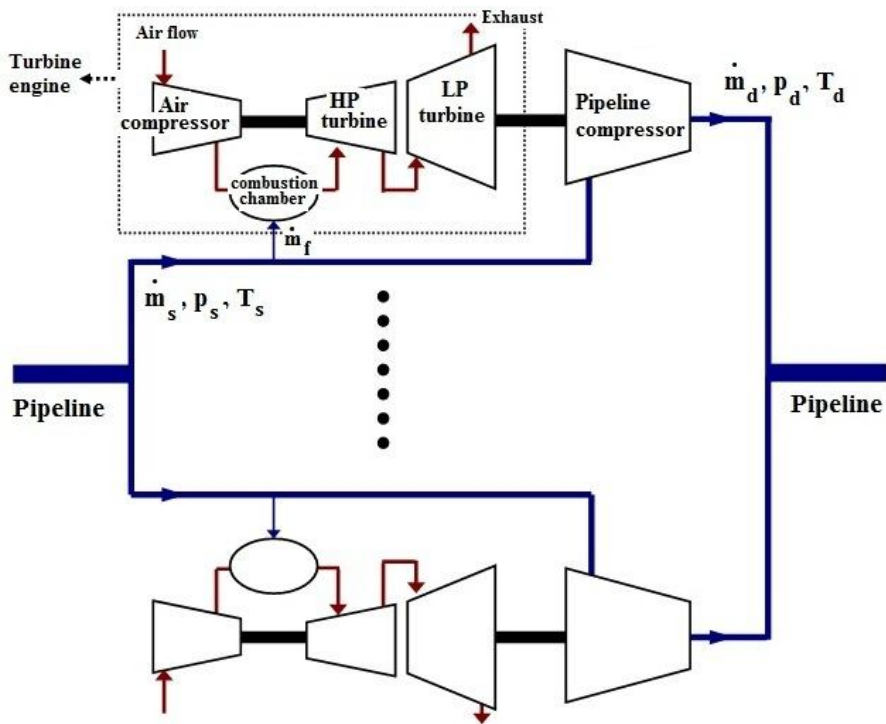
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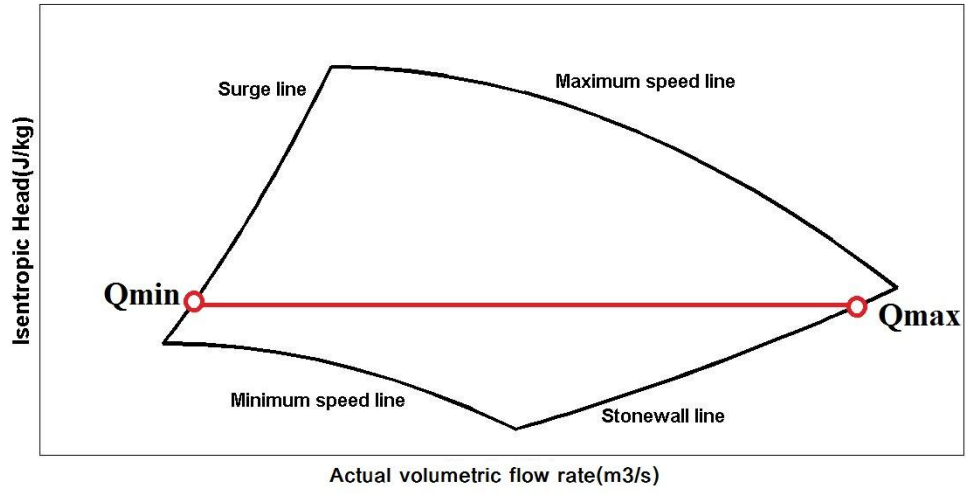


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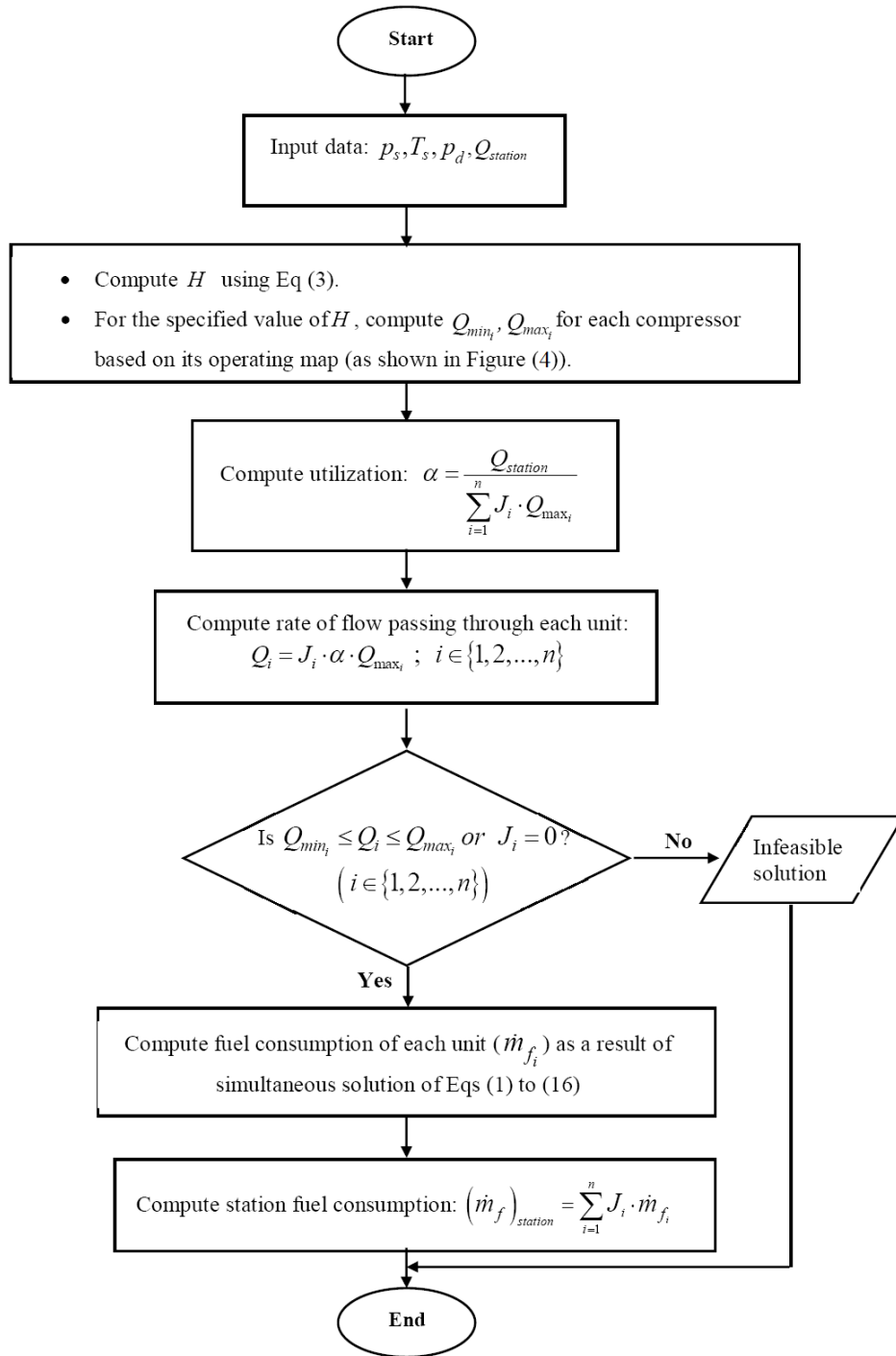


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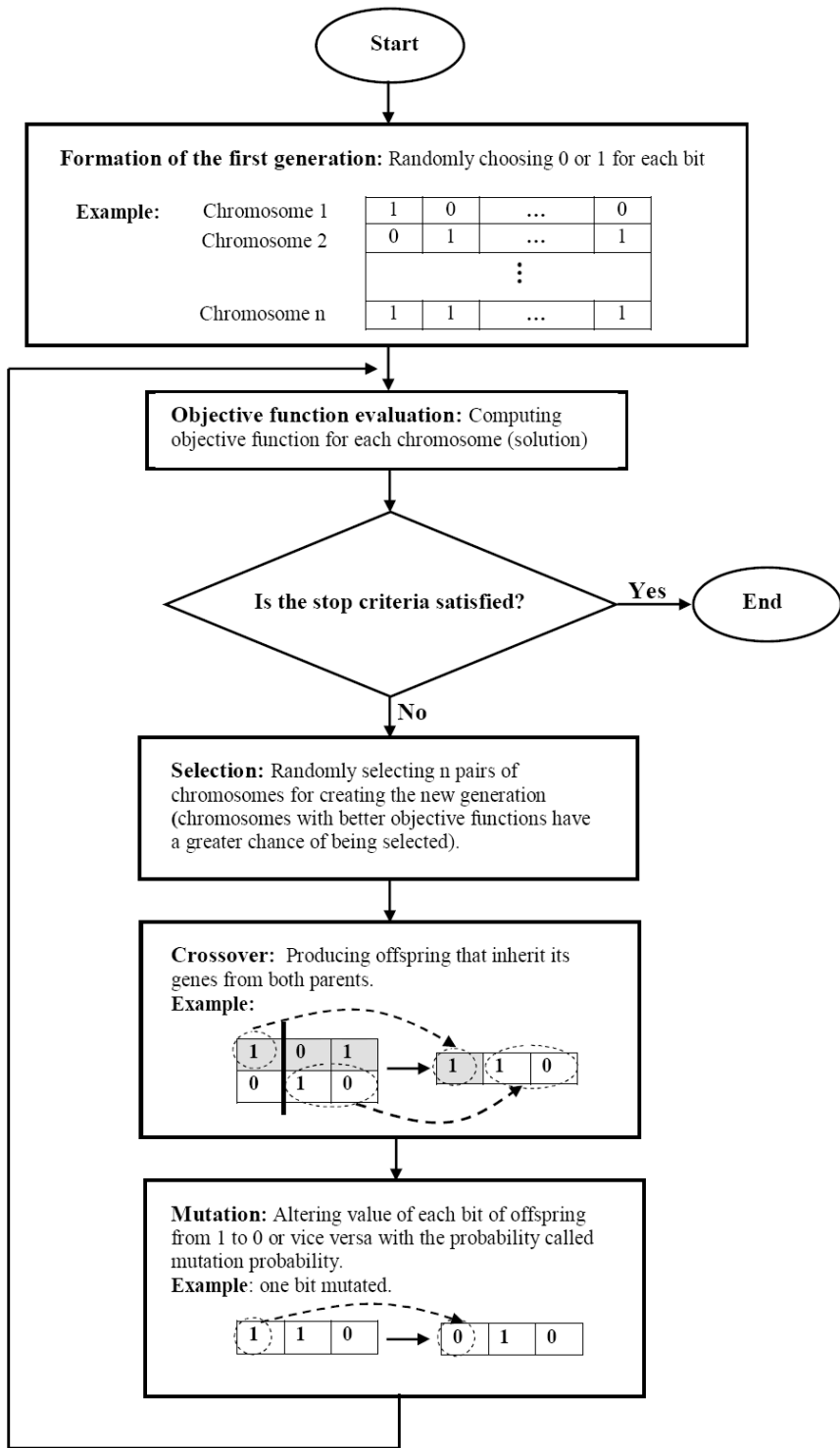
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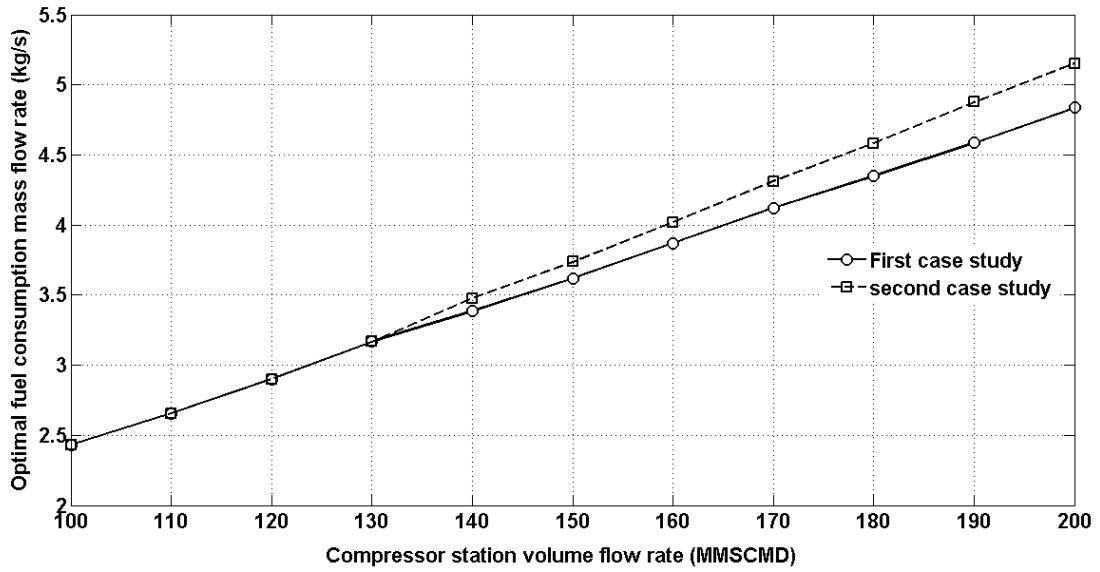


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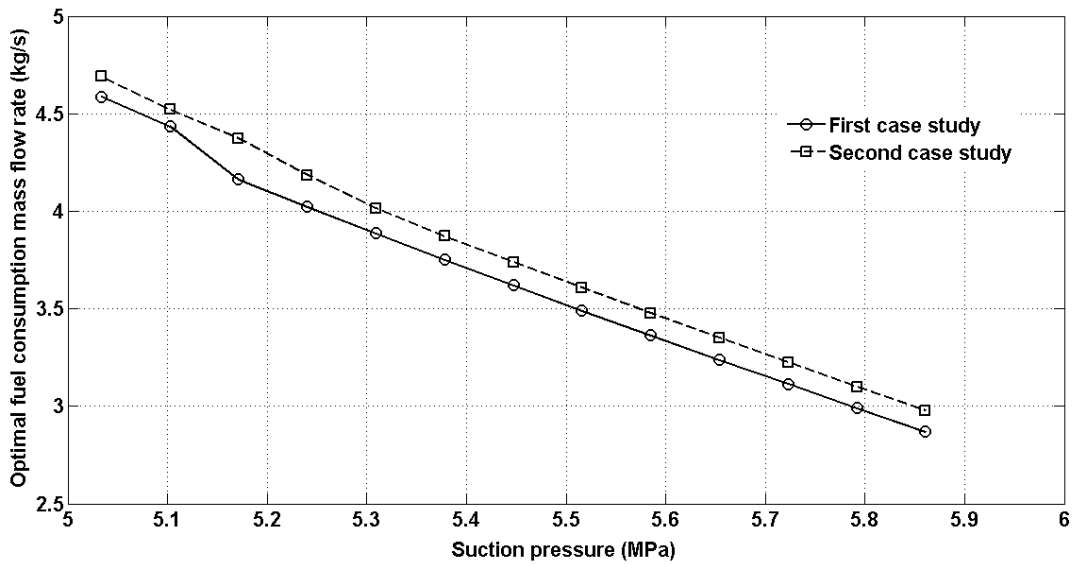


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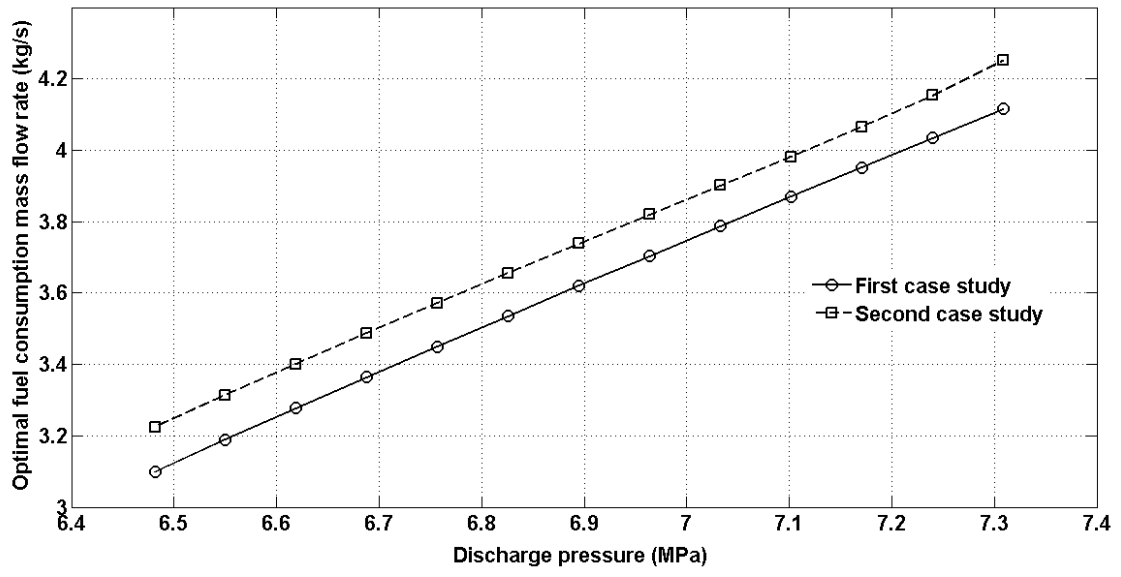
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